

FLTR 4.0 - automatic tool for blade flutter prediction in ANSYS CFX

KEY PURPOSES:

- Automation of blade flutter prediction in ANSYS CFX for turbines and compressors
- Complex unsteady CFD analysis with all geometric details
- Significant reduction of manual preprocessing and postprocessing 95%
- Reduction in simulation time more than 60%
- Comparability of flutter results same methodology



DESCRIPTION:

Design of the rotor blades for rotating machines is a complex procedure involving the study of the interaction between structural dynamics and aerodynamic forces – flutter analysis. Flutter analysis

involves several steps that are constantly repeating: modal aerodynamic analysis, work calculation and damping ratio evaluation. We would like to introduce our tool for the automation of flutter analysis in ANSYS CFX. The tool automates the generation of transient input files for CFD simulations and the evaluation of damping ratio from aerodynamic work done on blades. Are you performing flutter analysis on daily basis and would you like to significantly reduce time of your analyses? We can help you with our FLTR tool.





OTHER BENEFITS:

- Excellent support with long time flutter knowledge
- Methodology validated against an experimental measurement
- Efficient flutter results storage (S-shapes, Flutter maps)
- FLTR tool training with a practical tutorial (onsite/online)
- 3D unsteady viscous flow/structural analysis
- Stage geometry with cover and tie-boss
- Stator/Rotor unsteady interaction
- Complex mode shapes
- Mode shapes precheck
- All types of turbomachines



LSB WITH COULER THE THE BOSS





SOFTWARE/HARDWARE REQUIREMENTS

- Supported OS: Windows/Linux
- ANSYS 2019 R3
- Python 3.6

CONTACT:

Do not hesitate to contact us for more information

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or visit our web > <u>www.numsolution.cz</u>

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Select wo	rking directory			Action
Z:/progr/fltr/fltr-test			Generate CFX shapes	
Configuration	~			Generate def files
Change ANSYS root	Stator blades:	46	-	
W:/ANSYS Inc/v193	Rotor blades:	36	* *	Postprocess
Machine type	Displacement (mm):	0.1		
C Turbine	Periods to solve:	6	*	Generate def files
 Compressor 	Timesteps per period:	100	*	ocherate der mes
Interface model	Min. coefficient loops:	1	*	Check mesh
C Transient Rotor Stator	Max. coefficient loops:	5	*	Create animation
 Mixing-Plane 	Solver processes:	32	*	
🗹 Rotor tip gap interface	Alloc factor:	1.5		-Flutter map
	Periods to post:	1	*	Select csv file
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Ci	onfig file 'fltr.cfg' and direct	ories	were g	enerated
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